



1  
00:00:10,400 --> 00:00:08,720  
satellites have a have a common problem

2  
00:00:11,629 --> 00:00:10,410  
and that is they're always power starved

3  
00:00:13,580 --> 00:00:11,639  
no matter where you are there's not

4  
00:00:16,450 --> 00:00:13,590  
enough power and conventional solar

5  
00:00:18,980 --> 00:00:16,460  
power comes from rigid deployable arrays

6  
00:00:21,200 --> 00:00:18,990  
typically with cover glass they're

7  
00:00:23,570 --> 00:00:21,210  
somewhat fragile easily damaged during

8  
00:00:25,820 --> 00:00:23,580  
packaging and it's difficult to get a

9  
00:00:27,320 --> 00:00:25,830  
lot of power in a small spacecraft which

10  
00:00:29,630 --> 00:00:27,330  
is the current generation of cube SATs

11  
00:00:30,980 --> 00:00:29,640  
and ultra small spacecraft what we're

12  
00:00:33,860 --> 00:00:30,990  
doing in this test is we're

13  
00:00:36,560 --> 00:00:33,870

demonstrating the ability to take next

14

00:00:38,900 --> 00:00:36,570

generation very flexible solar arrays

15

00:00:41,750 --> 00:00:38,910

fold them package them into a very small

16

00:00:44,090 --> 00:00:41,760

volume and deploy them once they're in

17

00:00:45,890 --> 00:00:44,100

space so that you can get a large

18

00:00:54,670 --> 00:00:45,900

surface area and generate a lot of power

19

00:00:58,939 --> 00:00:57,350

the way this all got started was we were

20

00:01:02,299 --> 00:00:58,949

working with several companies on very

21

00:01:03,740 --> 00:01:02,309

thin film lightweight solar cells these

22

00:01:04,850 --> 00:01:03,750

are very different than the solar cells

23

00:01:07,460 --> 00:01:04,860

that are used in space applications

24

00:01:09,740 --> 00:01:07,470

today and they they're very flexible

25

00:01:12,290 --> 00:01:09,750

they're very light and they're mountable

26

00:01:15,080 --> 00:01:12,300

on on a variety of substrates including

27

00:01:16,280 --> 00:01:15,090

aluminum or captain and we were thinking

28

00:01:18,320 --> 00:01:16,290

about structures that we can mount them

29

00:01:19,670 --> 00:01:18,330

on and we realized that inflatable

30

00:01:21,530 --> 00:01:19,680

structures were were the perfect

31

00:01:24,680 --> 00:01:21,540

mounting structures for those and that

32

00:01:26,270 --> 00:01:24,690

they gave us a very rigid structure that

33

00:01:29,600 --> 00:01:26,280

would hold the lightweight array and

34

00:01:31,219 --> 00:01:29,610

would also provide very small storage

35

00:01:37,660 --> 00:01:31,229

volume for the array that you end up

36

00:01:43,210 --> 00:01:39,820

the unit you see who came from a concept

37

00:01:46,210 --> 00:01:43,220

study that we did for a micro set that

38

00:01:48,400 --> 00:01:46,220

would implement a telescope application

39

00:01:50,590 --> 00:01:48,410

and so we took the concept design that

40

00:01:51,610 --> 00:01:50,600

we had and actually built the test

41

00:01:54,400 --> 00:01:51,620

article out of it

42

00:01:56,080 --> 00:01:54,410

the silver substrate pieces that you see

43

00:01:57,850 --> 00:01:56,090

aren't print of the solar arrays but we

44

00:02:00,250 --> 00:01:57,860

had one actual solar cell that we

45

00:02:02,410 --> 00:02:00,260

attached to the pedal instrumented and

46

00:02:03,969 --> 00:02:02,420

we did flash tests of it in the

47

00:02:05,770 --> 00:02:03,979

beginning and we wrapped it up for

48

00:02:07,300 --> 00:02:05,780

deployment now we've done the deployment

49

00:02:09,219 --> 00:02:07,310

and you know do another test to assure

50

00:02:16,620 --> 00:02:09,229

that the deployment hasn't damaged it

51  
00:02:20,400 --> 00:02:18,480  
there are a lot of applications for this

52  
00:02:22,740 --> 00:02:20,410  
technology we envision that you could

53  
00:02:25,170 --> 00:02:22,750  
use this inflatable or deployables solar

54  
00:02:27,360 --> 00:02:25,180  
array system everything from cube SATs

55  
00:02:29,520 --> 00:02:27,370  
which are was a very small 10 centimeter

56  
00:02:31,560 --> 00:02:29,530  
by 10 centimeter spacecraft which are

57  
00:02:33,510 --> 00:02:31,570  
typically very very power limited

58  
00:02:36,030 --> 00:02:33,520  
something like this could be packaged in

59  
00:02:38,400 --> 00:02:36,040  
about 1/3 of the volume and inflate to

60  
00:02:40,800 --> 00:02:38,410  
something this size providing unheard of

61  
00:02:43,020 --> 00:02:40,810  
power for small SATs primarily in

62  
00:02:44,520 --> 00:02:43,030  
low-earth orbit you could also take this

63  
00:02:47,160 --> 00:02:44,530

technology and use it for human

64

00:02:49,560 --> 00:02:47,170

exploration in deep space and deploy

65

00:02:52,380 --> 00:02:49,570

incredibly large thanks size of this

66

00:03:09,020 --> 00:02:52,390

room or bigger arrays packaged into

67

00:03:14,070 --> 00:03:12,090

the next steps for the technology are to

68

00:03:16,470 --> 00:03:14,080

take this demonstration which we believe

69

00:03:19,559 --> 00:03:16,480

is shown of the viability of this at

70

00:03:21,570 --> 00:03:19,569

technology readiness level 4 and start